

Docket No. SA-522

Exhibit No. 7-Q

NATIONAL TRANSPORTATION SAFETY BOARD

Washington, D.C.

Airbus Vertical Tail High Loads for In Service Events

(6 Pages)



Loads

- **LE10 - Vertical tail high loads in service events**

Loads

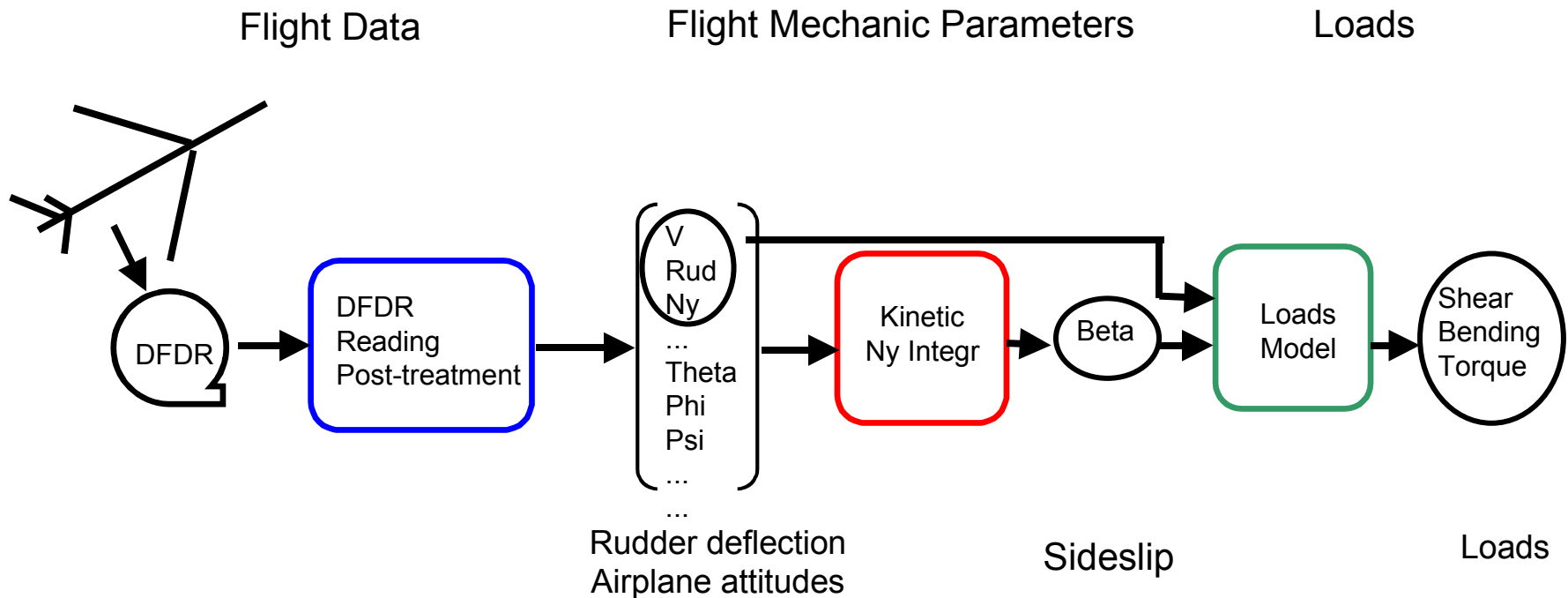
- **LE10 - Vertical tail high loads in service events**
 - Process analysis
 - Identification of events with potential high loads development on vertical tail.
 - Individual review of each event information/data
 - Classification with reference to four categories:
 - . Airplane “loss of control”
 - . Systems malfunctions
 - . Atmospheric disturbances (gust)
 - . Others (inadvertent engine reverse deployment in flight,..)
 - Establishment of the necessary data for loads analysis (time history).
 - . Controls deflection (rudder,...) and aircraft movement parameters (sidestep,...) using the “Kinetic/Ny integration” method.
 - Loads analysis
 - Decision on structural inspection (program, execution)
 - Review of structural inspection feed-back (finding).

— Loads

- LE10 - Vertical tail high loads in service events.

- Event analysis principle

- Kinetic/Ny Integration method used.
- Process applied



Loads

● LE10 - Vertical tail high loads in service events - A300-600

EVENT DATE	Event Family	Event	Speed (Vcas)	Config	Event	Crew Rudder input	Rudder doublet Yes/No	Ny (g)	Mx FinRoot	Detailed Inspection
NOV-01	OPERATION	A	250	Clean	successive Rudder doublets to TLU	yes	yes	0.38	1.96LL	Accident
MAY 97		B	190-230	Clean	Stall & Loss of control. Several Rudder doublets to TLU during recovery.	yes	yes	0.55 & 0.7	1.53LL (1st fully recorded doublet) Beyond UL estimate for subsequent doublets (DFDR data not recorded)	11 March 02 Local damages at rear RHS fin attachment.
MAY 89		C	250		Rudder jerk	yes	yes	0.33	1.11LL	16 March 02 No Finding
NOV 99	SYSTEM MALFUNCTION	D	220	Slats retract	Rudder trim runaway in climb at slat retract	yes	no	0.33	0.8LL	No
		E	225-290	Slats retract	Rudder trim runaway in climb at slats retract	yes	no	0.21	0.6LL	No
MAR 99		F	180-190	Slats out	Rudder oscillations in go around at AP disconnect	no	no	0.32	1.16LL	13 March 02 No Finding
NOV 98	OTHERS	T/R In Flight	G	300	Clean	Inadvertent thrust deployment		0.21	0.86LL	21 March 02 No finding
Gust		all cases reported/analysed show Ny < 0.3g, less lower than the one coming from design analysis								

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Loads

● LE10 - Vertical tail high loads in service events - A310

EVENT DATE	Event Family	Event	Speed (Vcas)	Config	Event	Crew Rudder input	Rudder doublet Yes/No	Ny (g)	Mx FinRoot	Inspection
FEB 91	OPERATION	Stall/ Loss of control	H	2	Missed approach followed by 3 successive stalls in go-around / loss of control with repetitive rudder movements.	yes	yes	0.36 0.69	1.55LL 1.35LL	Done 3 April 02 No Findings
SEP 94			I	3	Missed approach followed by stall in go-around.	yes	no	0.37	1.12LL	Done 26 March 02 No finding
NOV 99	SYSTEM MALFUNCTION	System failure	J	Clean	Rudder Trim runaway while AP engaged; lateral upset at AP disconnect	yes	yes	0.49	1.06LL	Done 28 March 02 No Findings
SEP 92			K	Clean	Rudder Trim runaway while AP engaged; lateral upset at AP disconnect followed by overspeed.	yes	no	0.32	0.8LL	No - Not recommended
Gust										all cases reported/analysed show Ny < 0.3g, less lower than the one coming from design analysis



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